

Characterization of Thermal & Mechanical Impact on Aluminum Honeycomb Structures

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Overview

1. About Me
2. Purpose
3. Destructive Analysis
4. NDE Methods
5. Future Work

About Me



Image courtesy of www.nationsonline.org

About Me

- Midwest → “Deep” south. SCIENCE!



Image courtesy of worldtouristmap.info



About Me



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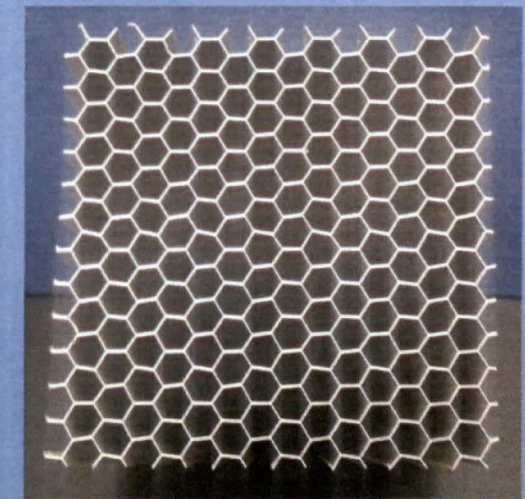
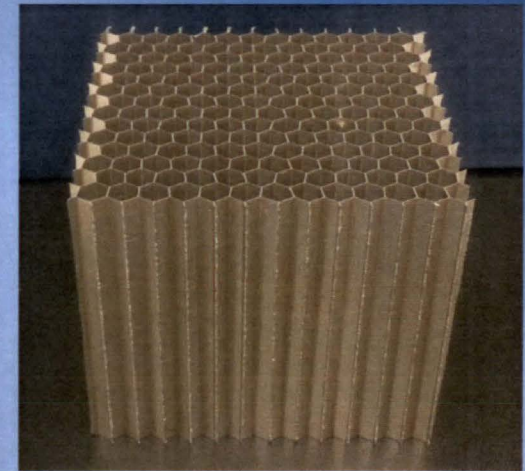


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What is the Purpose of This Research?

- Quantitative and qualitative data on Aluminum constructs
 - Energy absorption + Impact
 - Thermal performance
- NDE method development
 - Quality



Goals

1. Effectively measure damage point (volume/area)
2. Image internal compression damage non-destructively
3. Determine impact effects on cellular structure/strength
4. Compare thermal behavior of impacted vs. non-impacted specimens

Why so?

- Space = extreme ΔT environment (high temp down to cryogenic)
 - Habitation, etc.
- Mechanical and thermally Efficient
 - Honeycomb
 - 120° orientation = Surface area reduction
- High cost of fuel to travel in space
 - Aluminum → Durable, lightweight, affordable, tunable material
 - HIGH strength, LOW weight

Honeycomb Structure

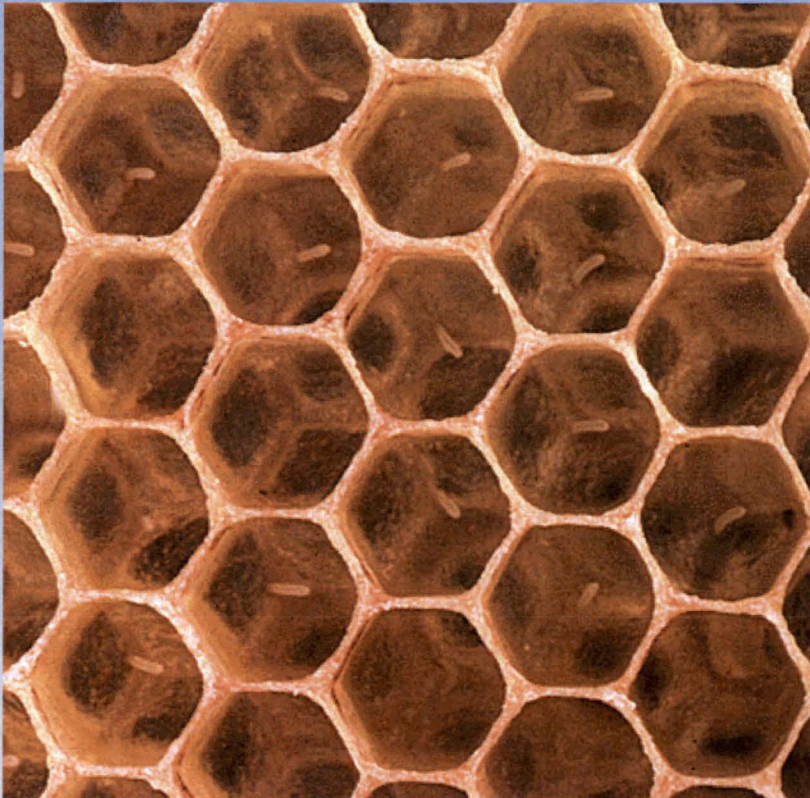


Image courtesy of
<http://http://phandroid.s3.amazonaws.com>

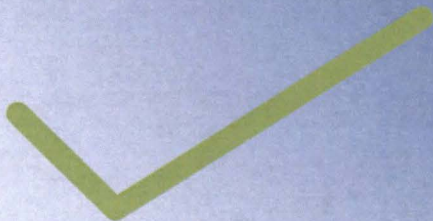
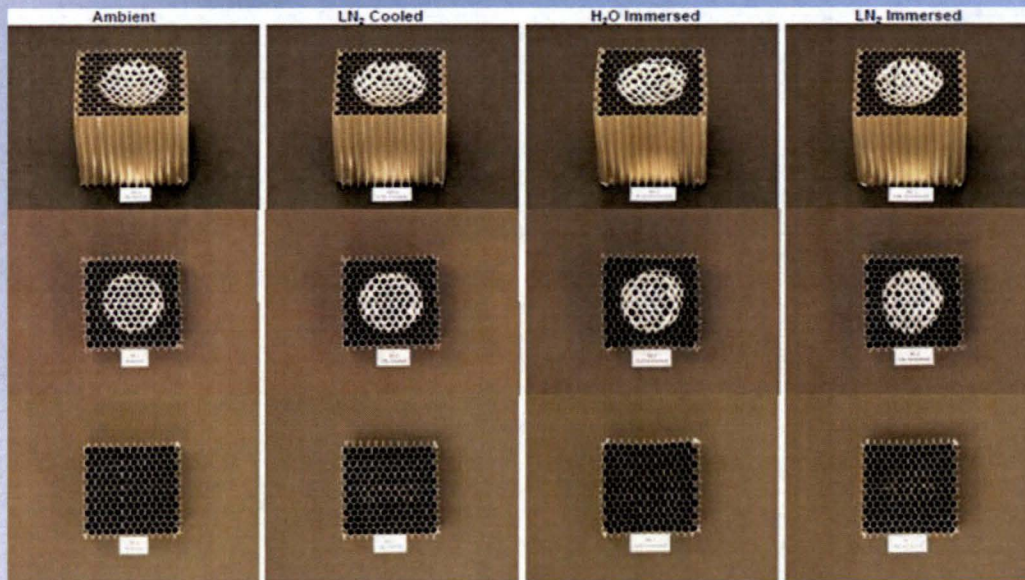
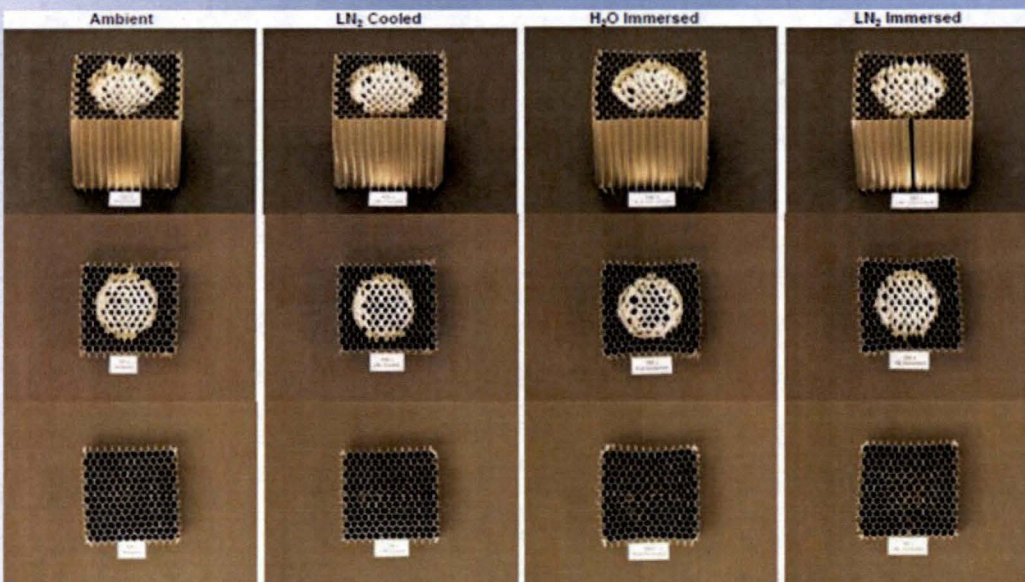


Image courtesy of <http://www.americansweets.co.uk>

Honeycomb Structure



50 J Impacted Samples



100 J Impacted Samples

Test Methods

Destructive

- Impact Test
- Compression Test

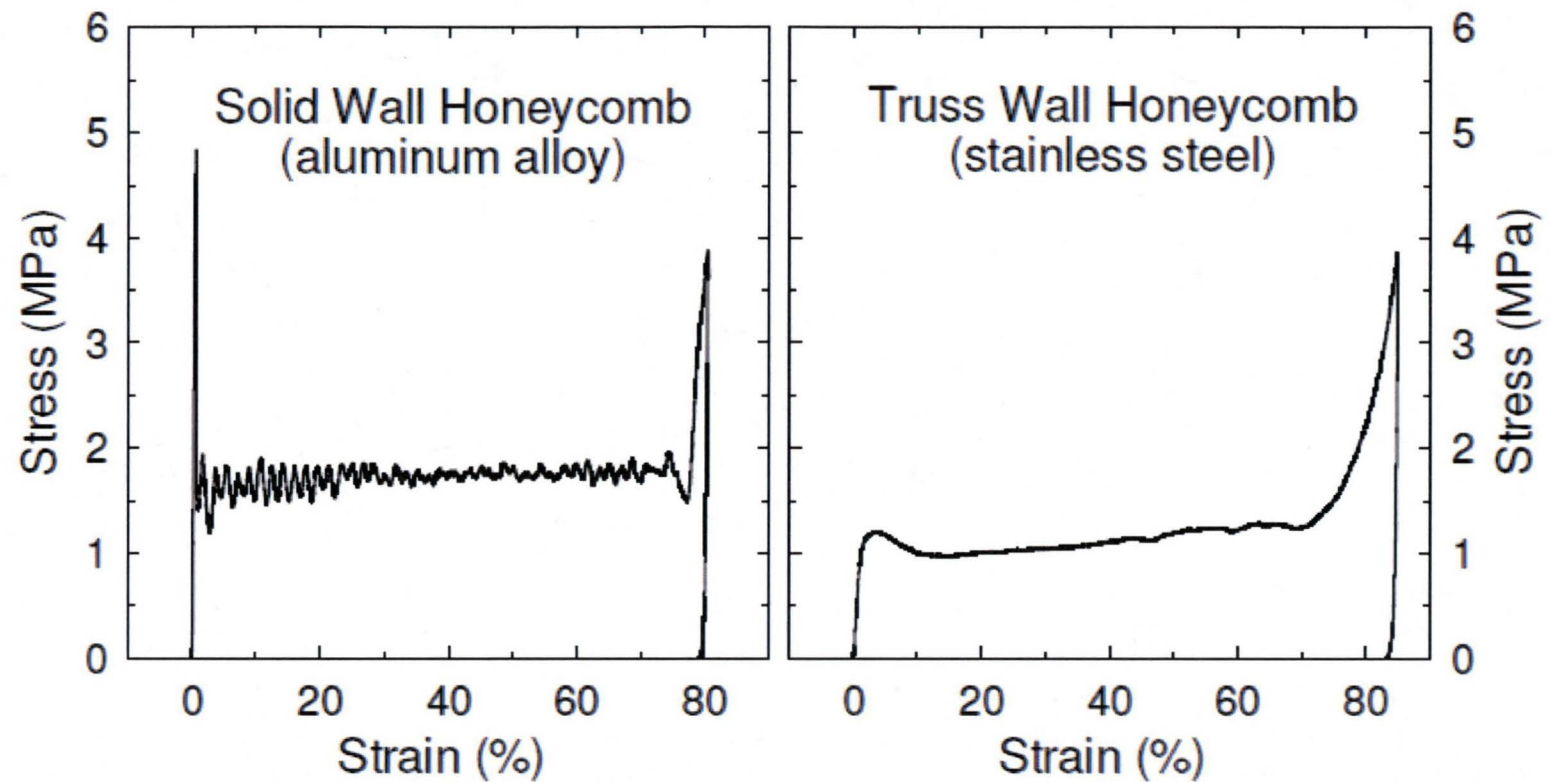
NDE: Mechanical/Thermal Testing

- Cryogenic Thermocouple (Heat flow)
- X-ray
 - CT Scanner
- CMM

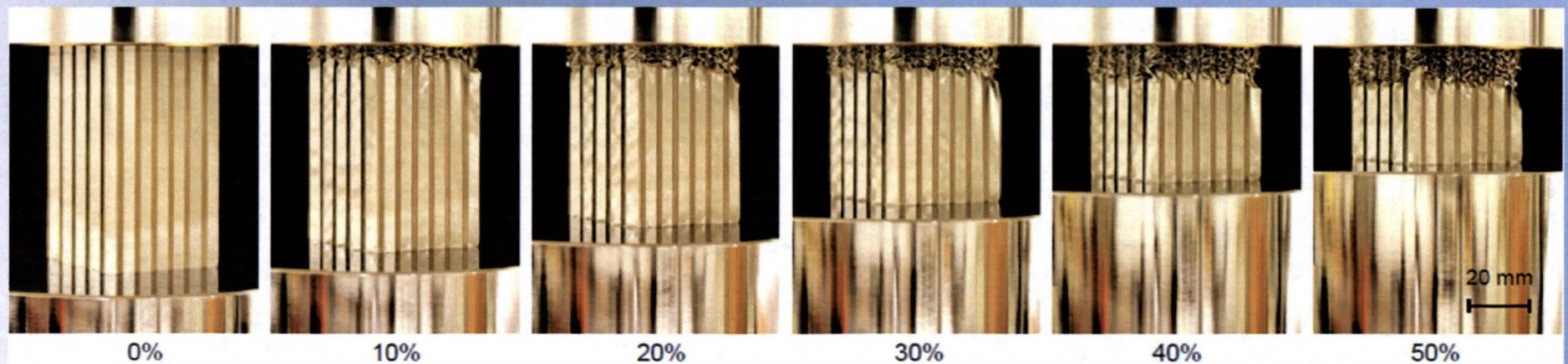
Destructive Evaluation

Compression and Impact Testing at Embry-Riddle
Aeronautical University

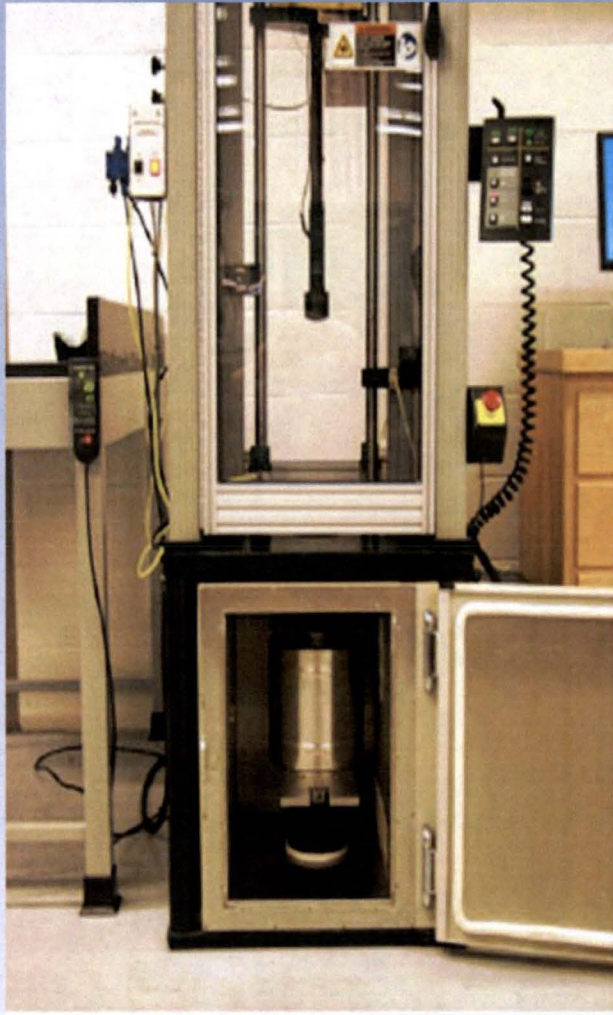
- 30,000 lbs compression
- Stress-Strain behavior
- 50 J Impact
- 100 J Impact
- 200 J + 300 J Impact



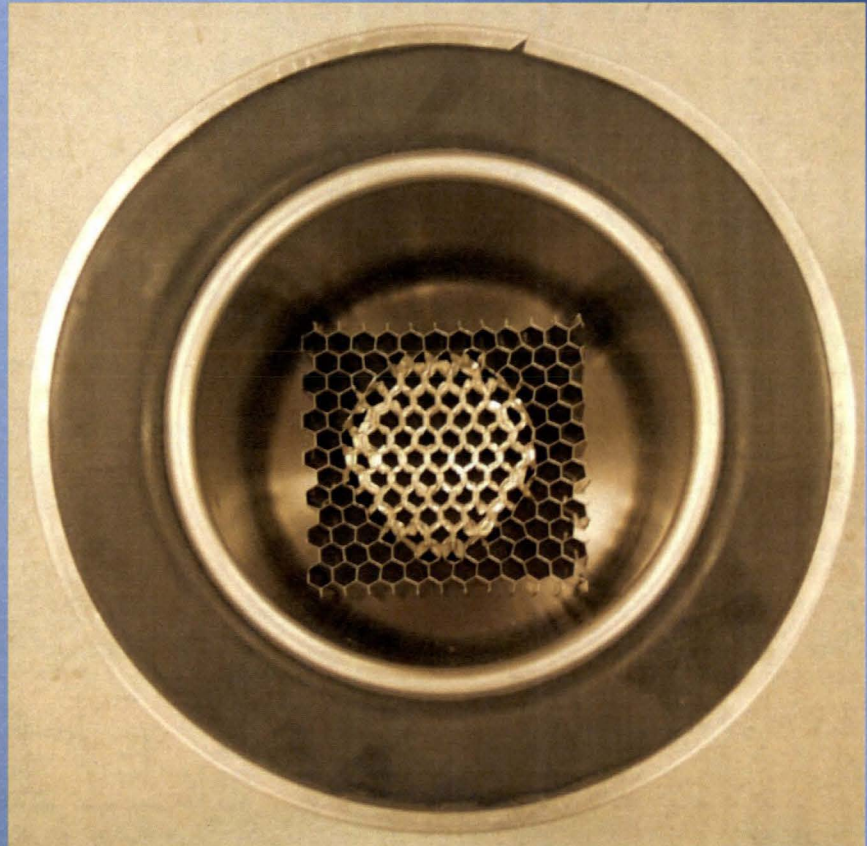
Stress-strain behavior for crushed solid wall (left) and truss wall (right) honeycombs



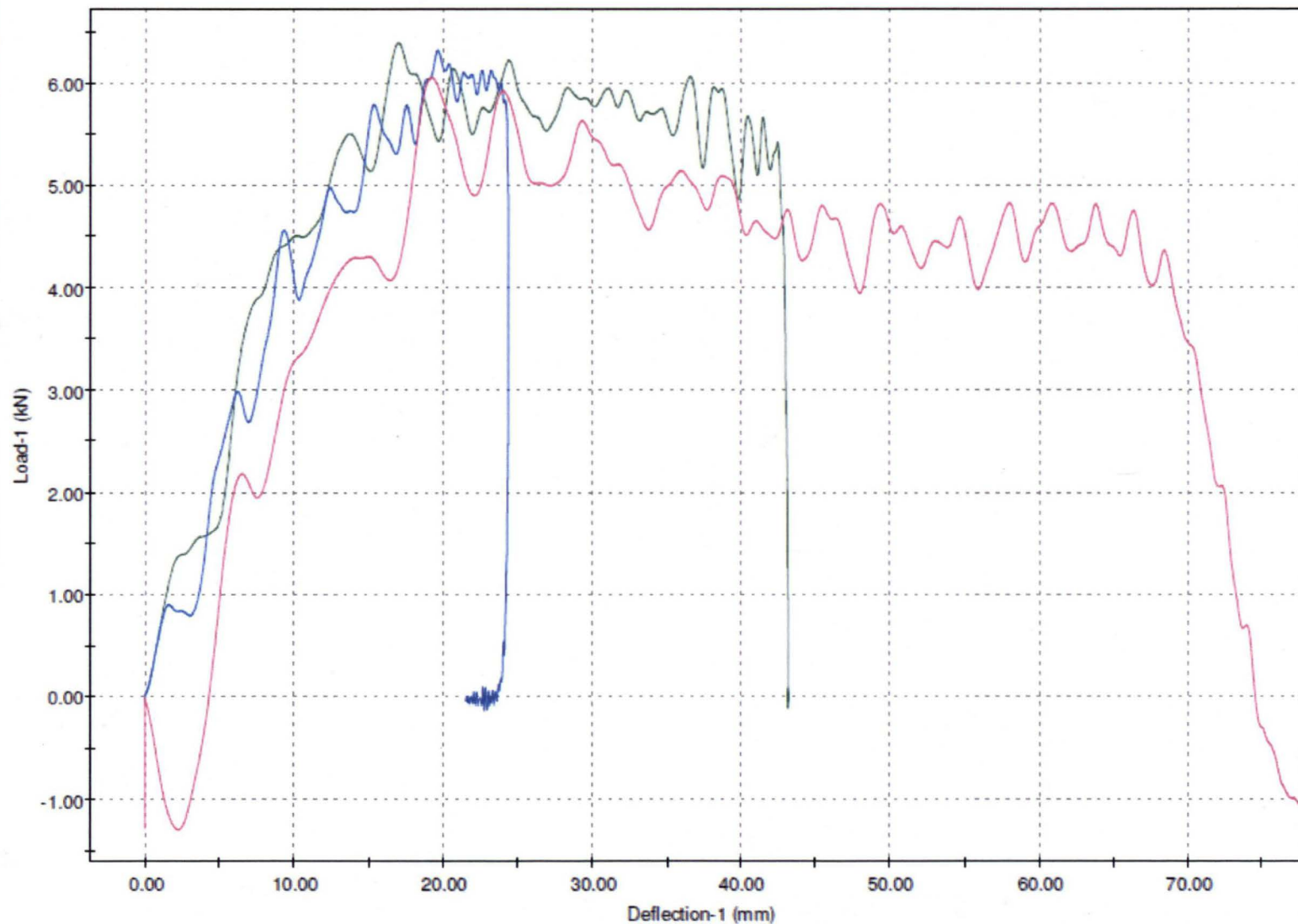
Impact Equipment



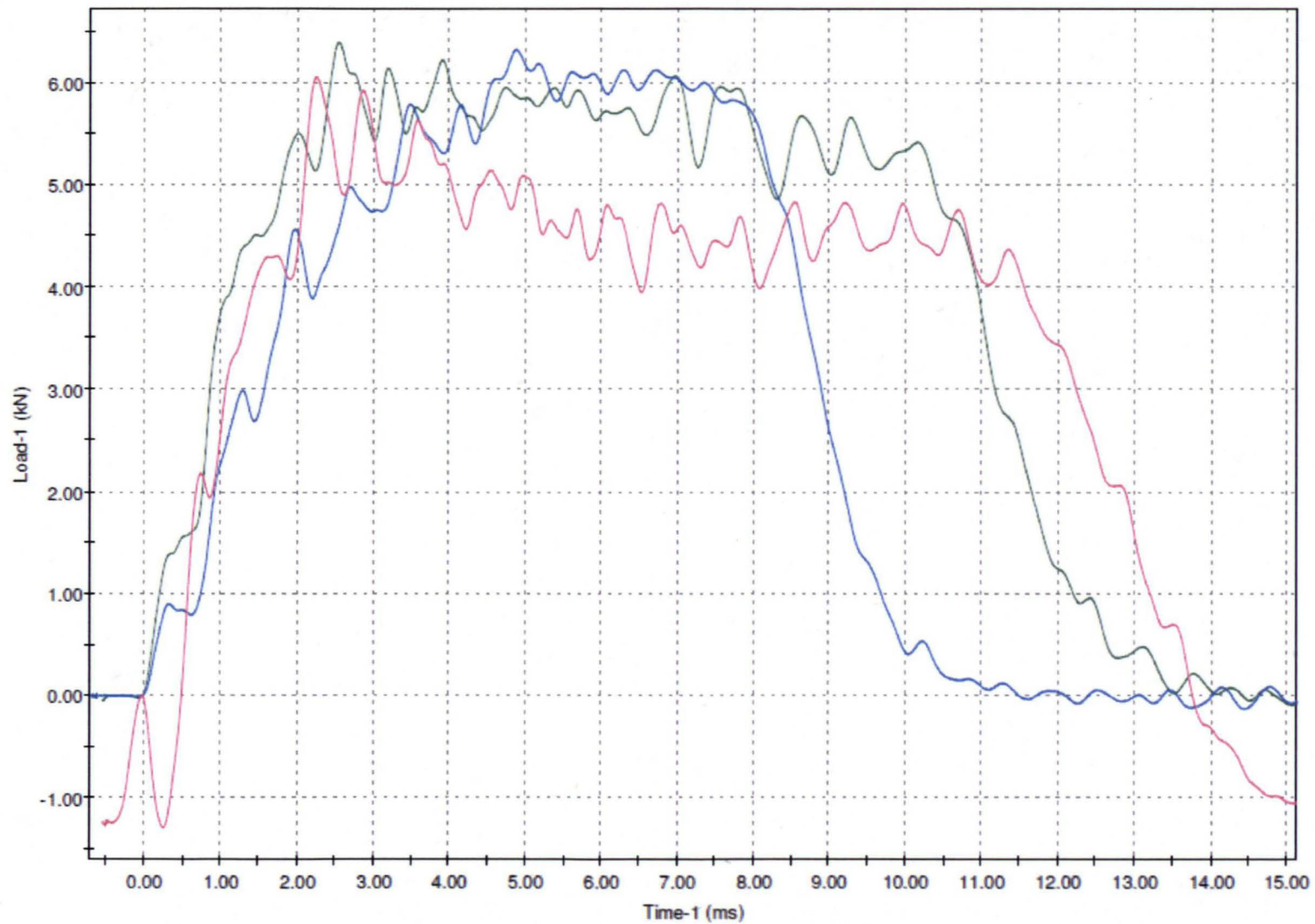
Instron 9250 Impact System



Force vs. Displacement, 100 J-300 J

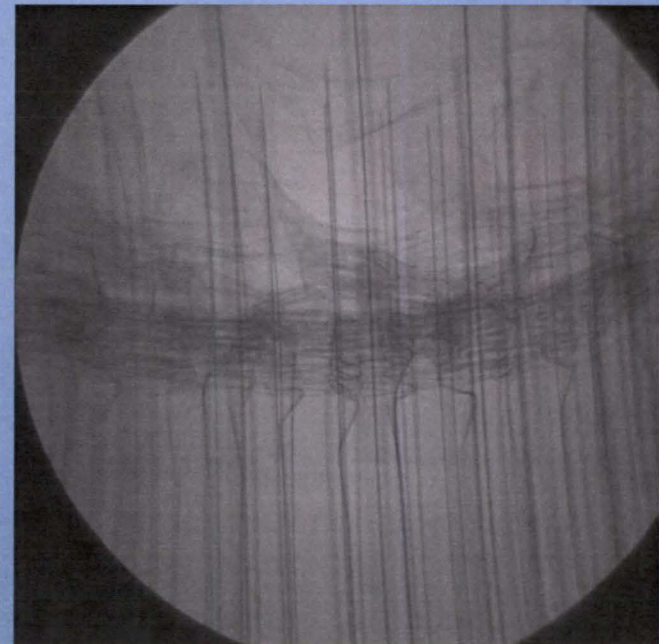
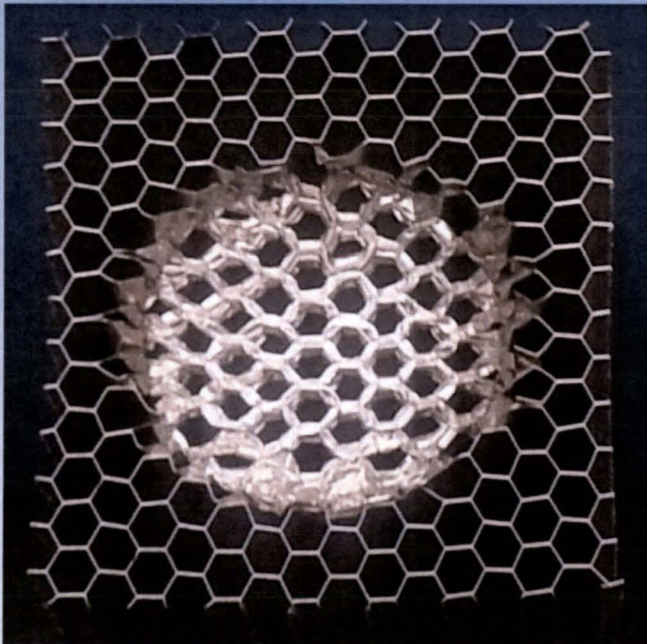


Force vs. Time, 100 J-300 J



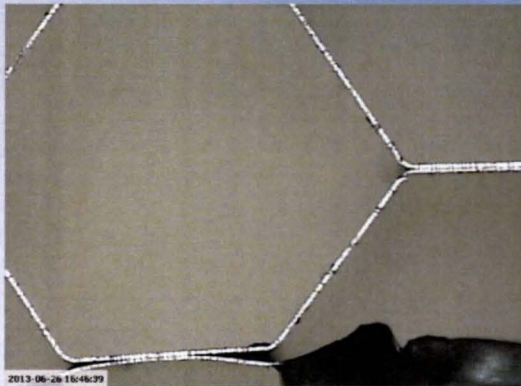
X-Ray Evaluation

- Beam and applicability
- CT Scanner
 - CAT Scan
 - Cross-Sectioning

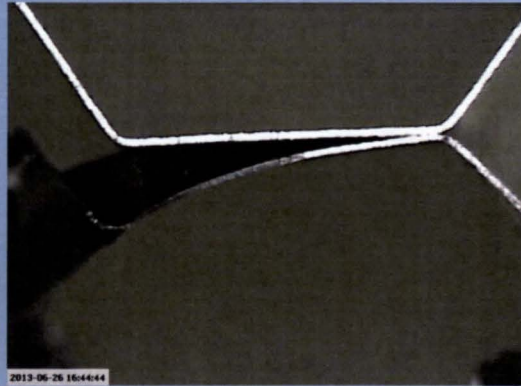


Metrology

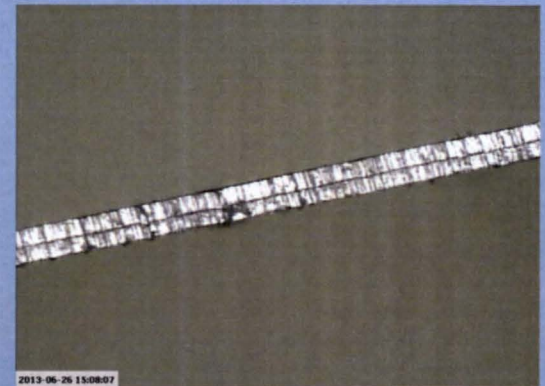
- Micro-Vu Excel 654 UC
 - Cell wall



20x



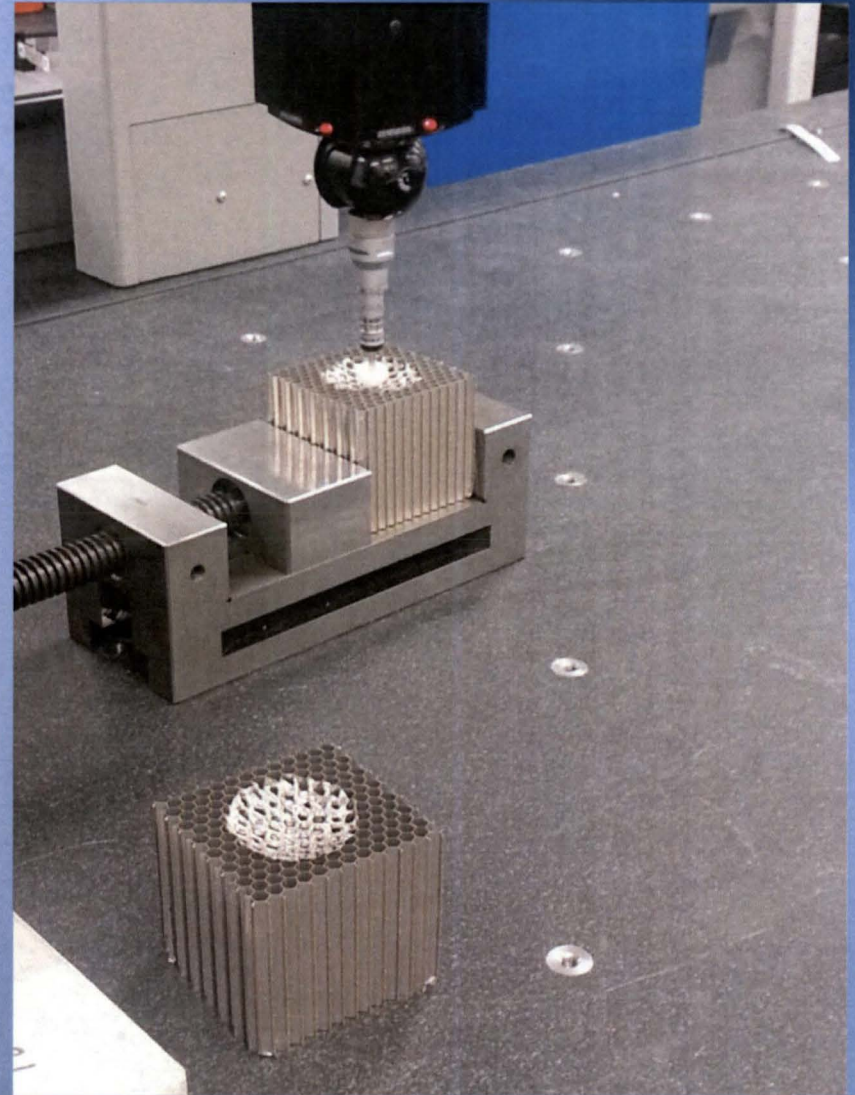
30x



82x

Metrology

- Micro-Vu Excel 654 UC
 - Cell wall
- Brown & Sharp Global Image 9128
 - Volumetric Analysis



Metrology

	Flatness (inches)	Depth 1 (inches)	Depth 2 (inches)	Depth 3 (inches)
Ambient (50 J)	0.0029	0.3736	0.3768	0.3754
LN₂ Immersed (50 J)	0.0019	0.3621	0.3617	0.3567
LN₂-cooled (50 J)	0.0018	0.3634	0.3646	0.3638
H₂O-Immersed (50 J)	0.0018	0.3489	0.3491	0.3481
Ambient (100 J)	0.0015	0.7230	0.7213	0.7257
LN₂ Immersed (100 J)	0.0027	0.6036	0.5973	0.5945
LN₂-cooled (100 J)	0.0028	0.5674	0.5689	0.5689
H₂O-Immersed (100 J)	0.0099	0.5831	0.5849	0.5841

Cryogenic Testing

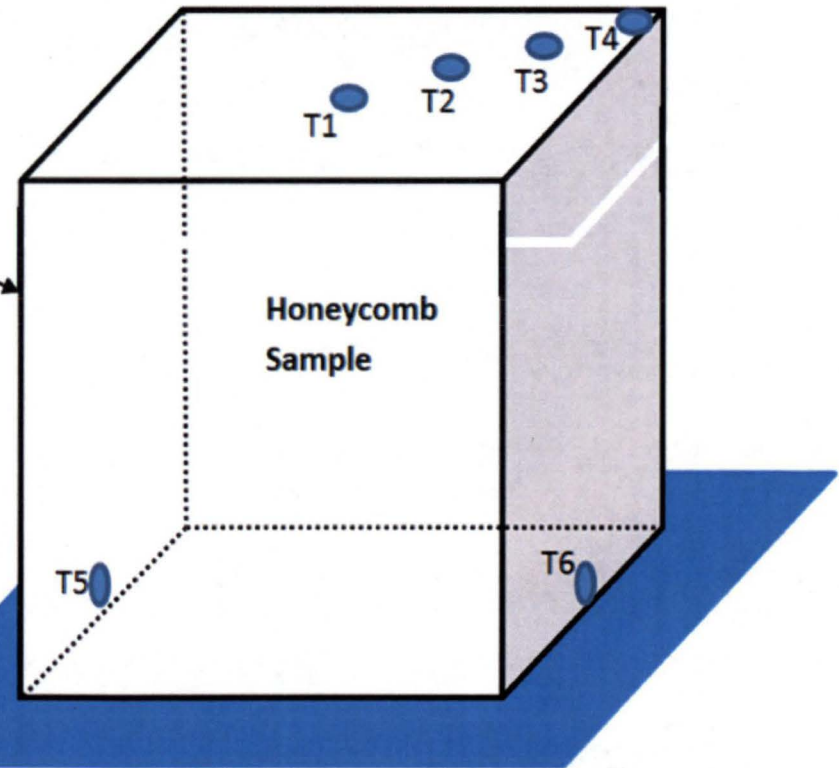
- Thermocouple purpose
- Test Theory

Heat
Gradient

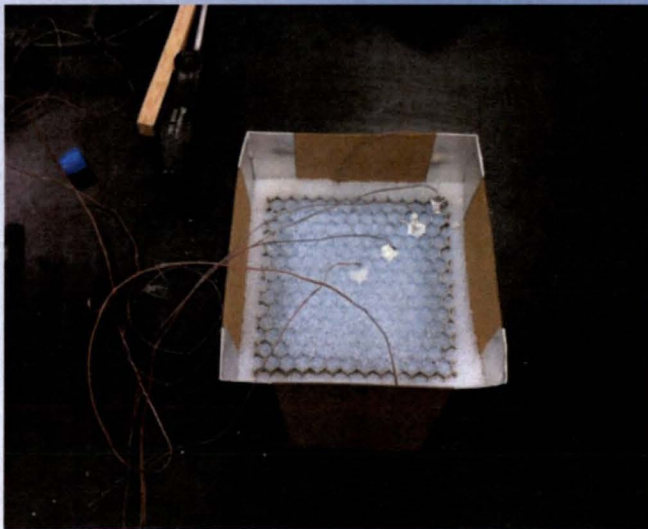


$T_0 = \text{ambient}$
(293 K @ $t=0$ sec)

Cold
Boundary

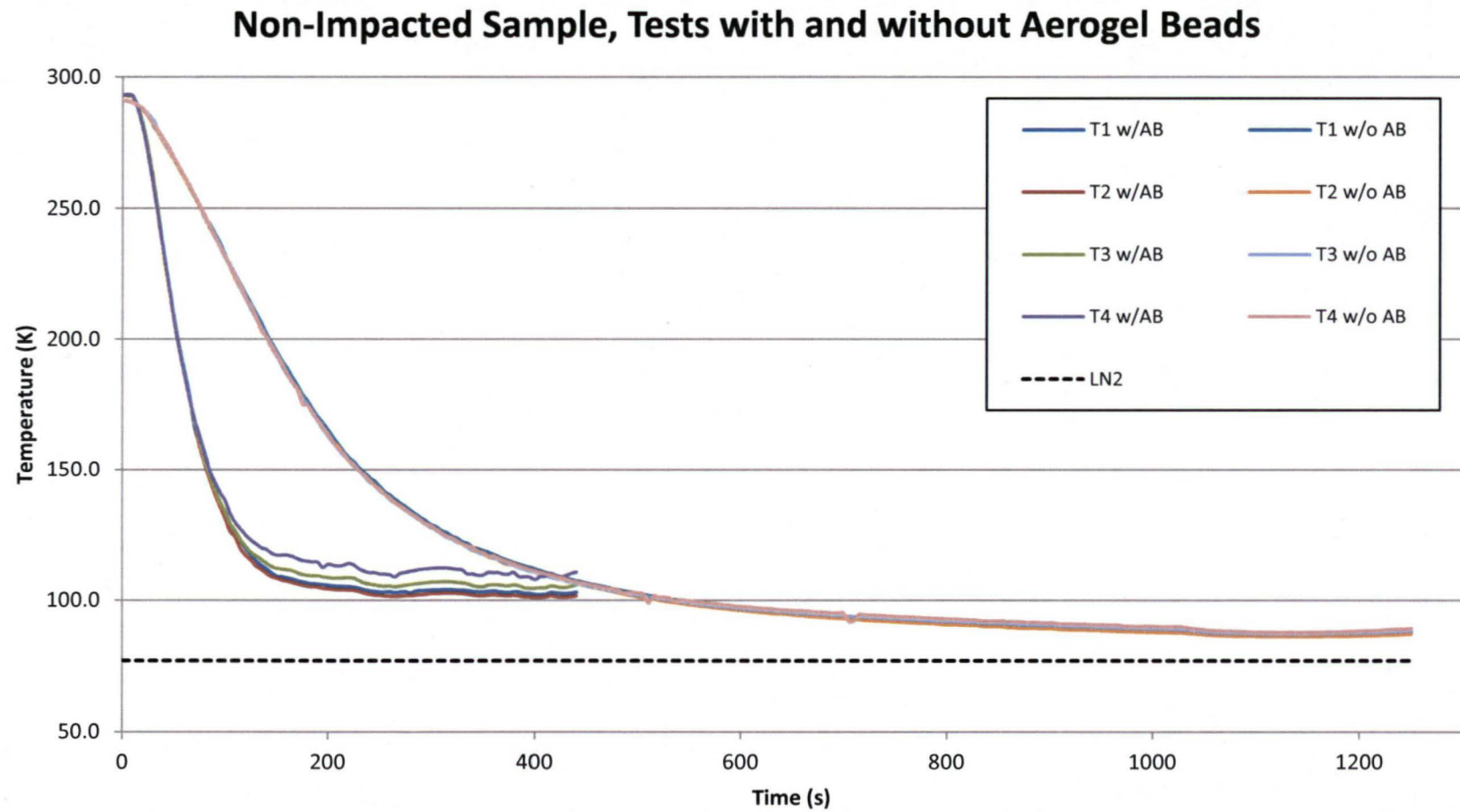


Cryogenic Testing



	<u>Test #1</u> w/o Insulation	<u>Test #2</u> With Insulation
Initial Sample Temp.	293.2 K	291.1 K
Avg. Steady State (SS) Temp.	101.8 K	86.1 K
Approx. Time to Reach SS Temp.	250 sec	1105 sec
Avg. ΔT Between Thermocouples	3.3 K	0.9 K
ΔT Between T1 and T4	7.2 K	0.7 K

Cryogenic Testing



Future Work

- Microscope → Adhesive
- High(er) impact testing
- High(er) impact cryogenic testing
 - More instrumentation, and refined test apparatus

Conclusions

- Impact Testing + Compression Testing
- X-Ray technology
 - CT Scanner
- Metrology
- Thermocouple
- Microscope
- Future:
 - White light scanner
 - Blue Light scanner
 - Flash thermography
 - Shearography

Acknowledgements

Many thanks to NASA, Kennedy Space Center for making this research effort possible.